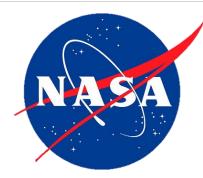
National Aeronautics and Space Administration





X-57 "Maxwell" High-Lift Propeller Testing and Model Development

2021 AIAA Aviation Forum: TFS-16/VSTOL-06

AIAA-2021-3193

BRANDON LITHERLAND NASA Langley Research Center Aeronautics Systems Analysis Branch NICHOLAS BORER NASA Langley Research Center Aeronautics Systems Analysis Branch NIKOLAS ZAWODNY NASA Langley Research Center Aeroacoustics Branch

brandon.l.litherland@nasa.gov nicholas.k.borer@nasa.gov nikolas.s.zawodny@nasa.gov

Wednesday, August 4, 2021



X-57 Introduction

NASA

The X-57 "Maxwell", NASA's next manned X-plane, incorporates relatively small diameter, folding, electricallydriven propellers along the wing leading-edge that provide lift augmentation at low speeds.

14 CFR § 23.2110(b)

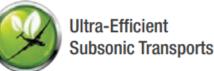
X-57 Mod IV flights will be the backbone of a means of compliance for low-speed handling for advanced air vehicles in response to recent changes to FAA airworthiness rules.

Credit: NAS/



Introduction and Background

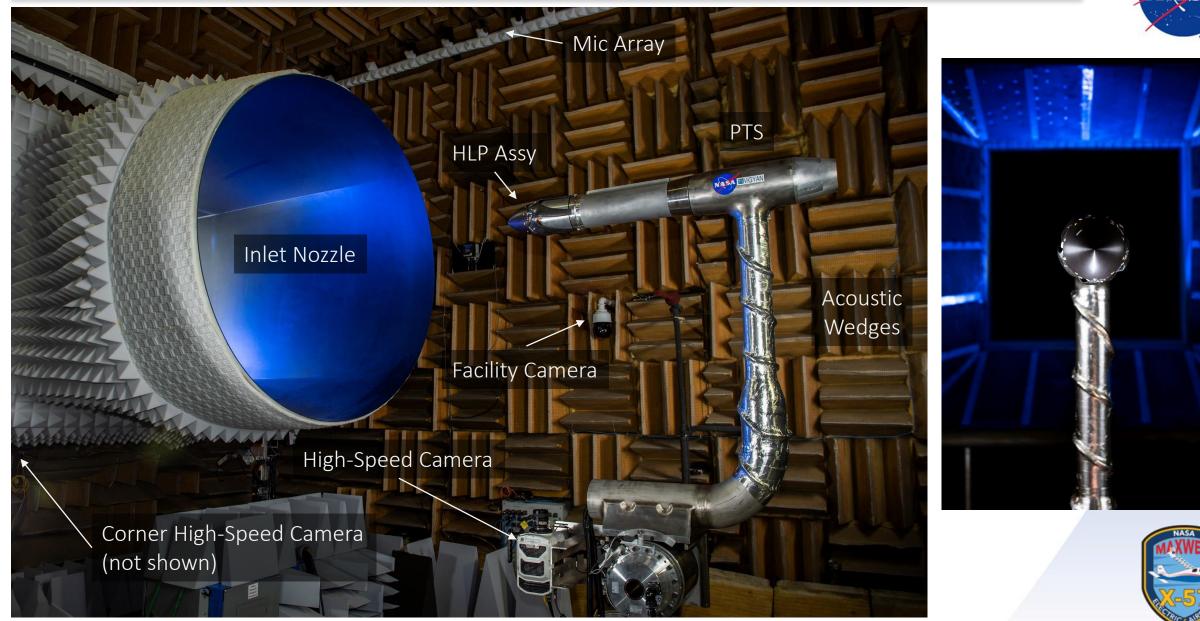
- Novel propeller designs for advanced air mobility (AAM) vehicles can have traits that make accurate performance prediction challenging using traditional methods.
 - Directly supports NASA ARMD ST3 & ST4:





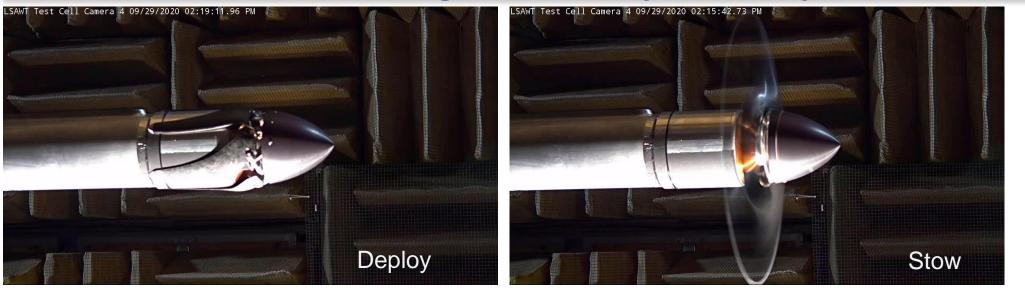
- Safe, Quiet, and Affordable Vertical Lift Air Vehicles
- Testing, model maturation, and standards development ensure U.S. leadership in AAM and the sustainability of National airspace and aviation.
- X-57 and Revolutionary Vertical Lift Technologies (RVLT) Project collaborated to test two full-scale high-lift propellers (HLPs) in the Low Speed Aeroacoustic Wind Tunnel (LSAWT) at NASA Langley Research Center.
 - Gathered performance and acoustic data to support model validation and improvements.
 - Captured high-speed video to evaluate propeller stability, dynamics, and blade position.
 - Performed shakedown test of RVLT's new Propeller Test Stand (PTS).
- LSAWT test helped to qualify propeller design and quantify performance and noise data, a critical piece of getting these unique propellers in the air.

X-57 Maxwell High-Lift Propeller Test Setup



brandon.l.litherland@nasa.gov

X-57 Maxwell High-Lift Propeller Operation



The HLP demonstrated smooth and stable behavior throughout the test.



brandon.l.litherland@nasa.gov





5

0 KT Deploy

High-speed view from bottom

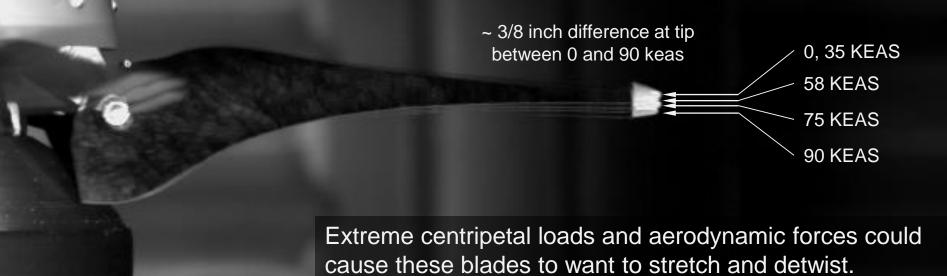
(1000 fps)

High-Lift Propeller Test High-Speed Imagery

Freestream speed effects at 4800 RPM

Target Screw Head

Multiple frames from high-speed capture are overlaid to create a digital "multiple exposure" image. Aligning these frames enables the examination of blade deflection and orientation at set conditions.

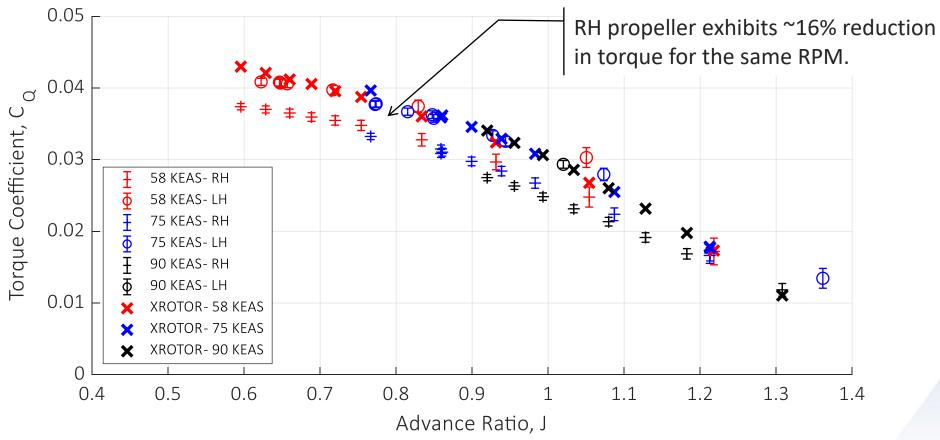


X-57

brandon.l.litherland@nasa.gov

X-57 Maxwell High-Lift Propeller Test Results

- > The LH propeller tracked with XROTOR model predictions remarkably well.
- Right-hand (RH) prop shows reduced performance compared to the left-hand (LH) prop.
- RH blades ~5% lower weight than LH blades. RH blades may partially unload at higher RPM.

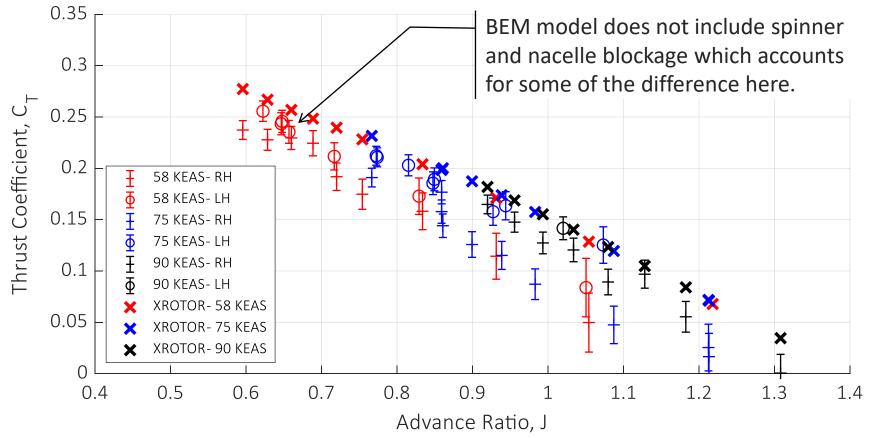


This material is a work of the U.S. Government and is not subject to copyright protection in the United States.

8

X-57 Maxwell High-Lift Propeller Test Results

- > Thrust measurements varied considerably due to PTS load cell issues.
 - Load-cell cross-loading and heat soaking caused drift in data.
 - Lower CT values show increased error in part due to larger relative error to thrust.
- > Fortunately, thrust data is more reliable at lower J where HLP is expected to operate.



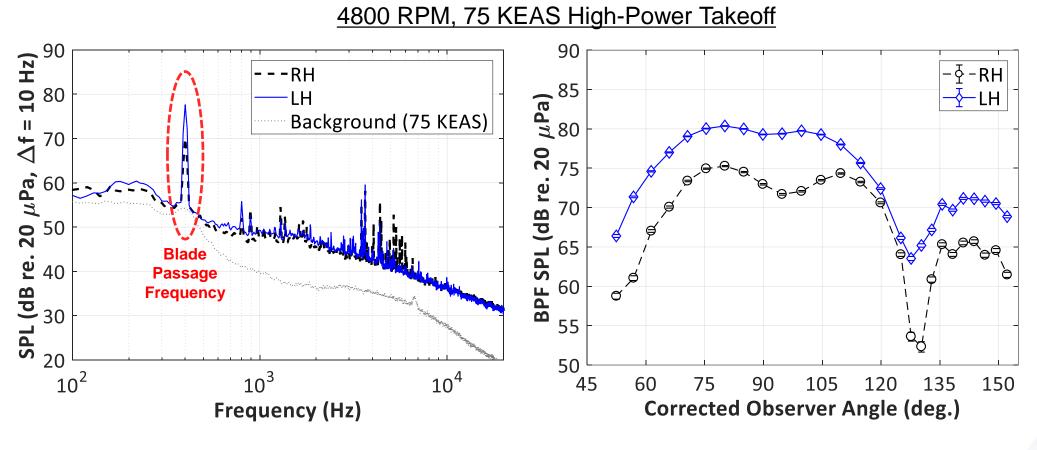
This material is a work of the U.S. Government and is not subject to copyright protection in the United States.



9

Test Results (contd.)

- Left-handed propeller found to exhibit higher fundamental tonal acoustic levels.
- Corroborated by higher torque levels and predictions revealing HLP to be dominated by aerodynamic loading noise.





X-57 High-Lift Propeller Test Impacts & Future Work

- Provided significant benefit to X-57 Maxwell and AAM vehicle studies.
 - Verified operation of a full-scale prototype of the flight propeller (~15 hours runtime).
 - HLP blade-element momentum (XROTOR) model predicted propeller performance well.
 - Obtained experimental data to improve modeled performance and dynamics.
 - No instability or excessive vibration observed under all operating conditions.
 - Generated practices for static balancing of a 5-bladed, folding, spring-loaded propeller.
- > Knowledge transfer to public, other Agencies, and industry partners.
 - X-57 Technical Publications Page
 - https://www.nasa.gov/aeroresearch/X-57/technical/index.html
 - Advances airworthiness standards and provides novel acoustic reference platform for comparison of AAM technologies.
- Future Work
 - X-57 folding HLP dynamics test to establish damping at the blade hinge.
 - Follow-up HLP performance test with similarly weighted blades from the same batch.



Acknowledgments



Transformational Aeronautics Concepts Program

- Convergent Aeronautics Solutions Project -

– Transformational Tools and Technologies Project –

Integrated Aviation Systems Program

Flight Demonstrations and Capabilities Project –
– X-57 Maxwell Subproject – –

High-Lift Propeller CAD/Fabrication

– Empirical Systems Aerospace, Inc. –

Advanced Air Vehicle Program

Revolutionary Vertical Lift Technology Project –

Propeller Test Stand Design/Build

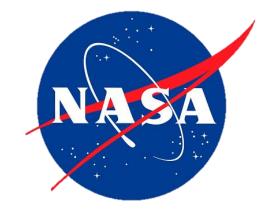
– ViGYAN, Inc. –

We also thank the LaRC LSAWT personnel for supporting this test and the LaRC Media Solutions Branch personnel for photography and high-speed imagery support.



All images in this document are credited to NASA.

brandon.l.litherland@nasa.gov



Questions?

